

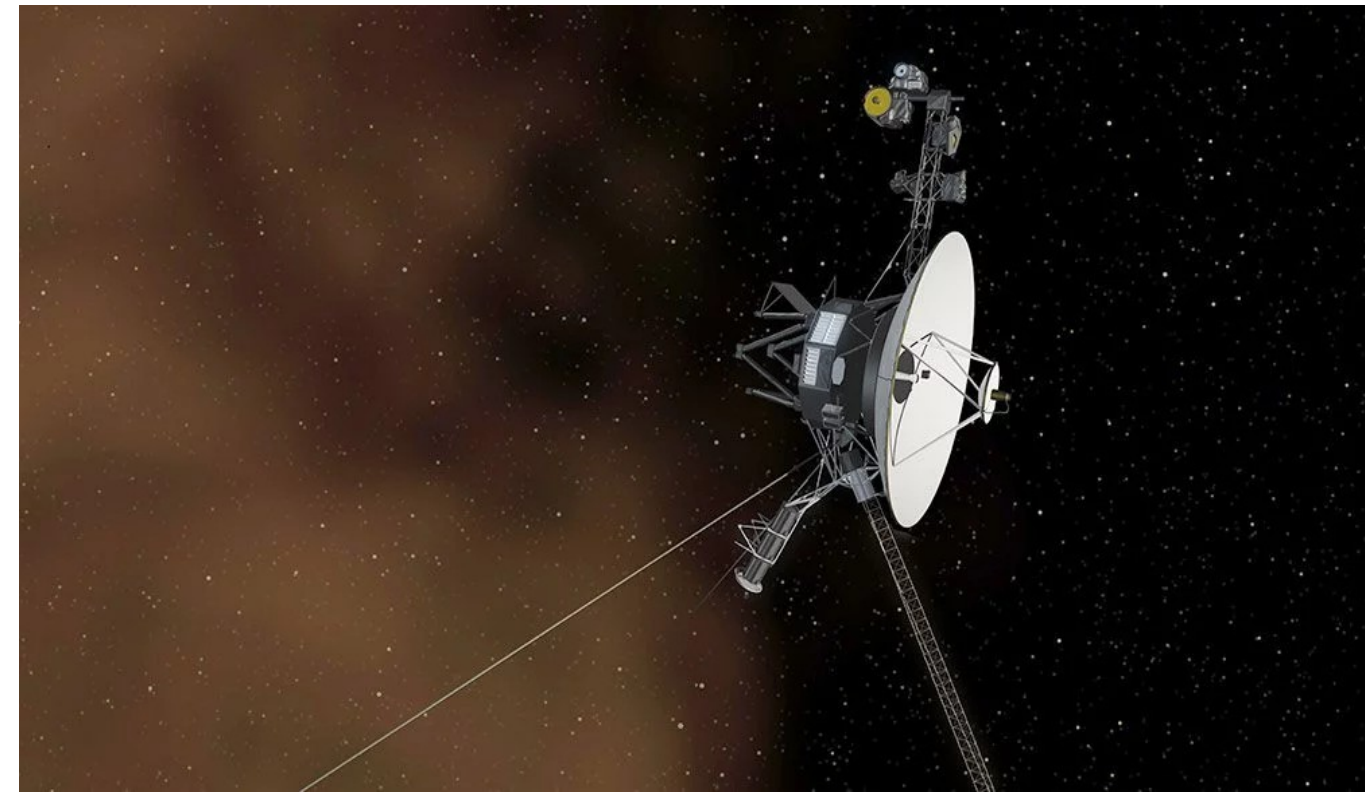
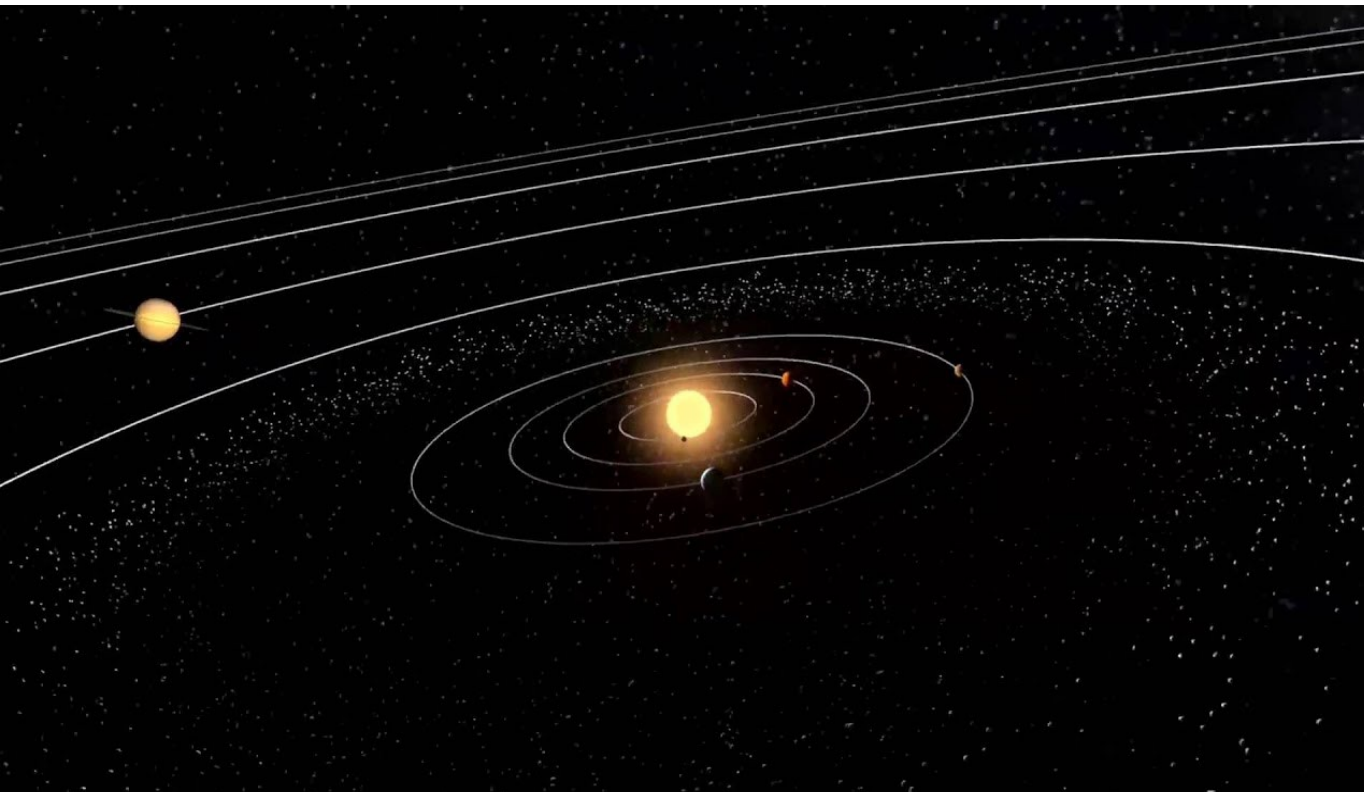
Simulation of an N-Body System and Swing-by Maneuver

Aaron Nagel – aaron.nagel@stud.uni-goettingen.de

Yannik Feldner – yannik.feldner@stud.uni-goettingen.de

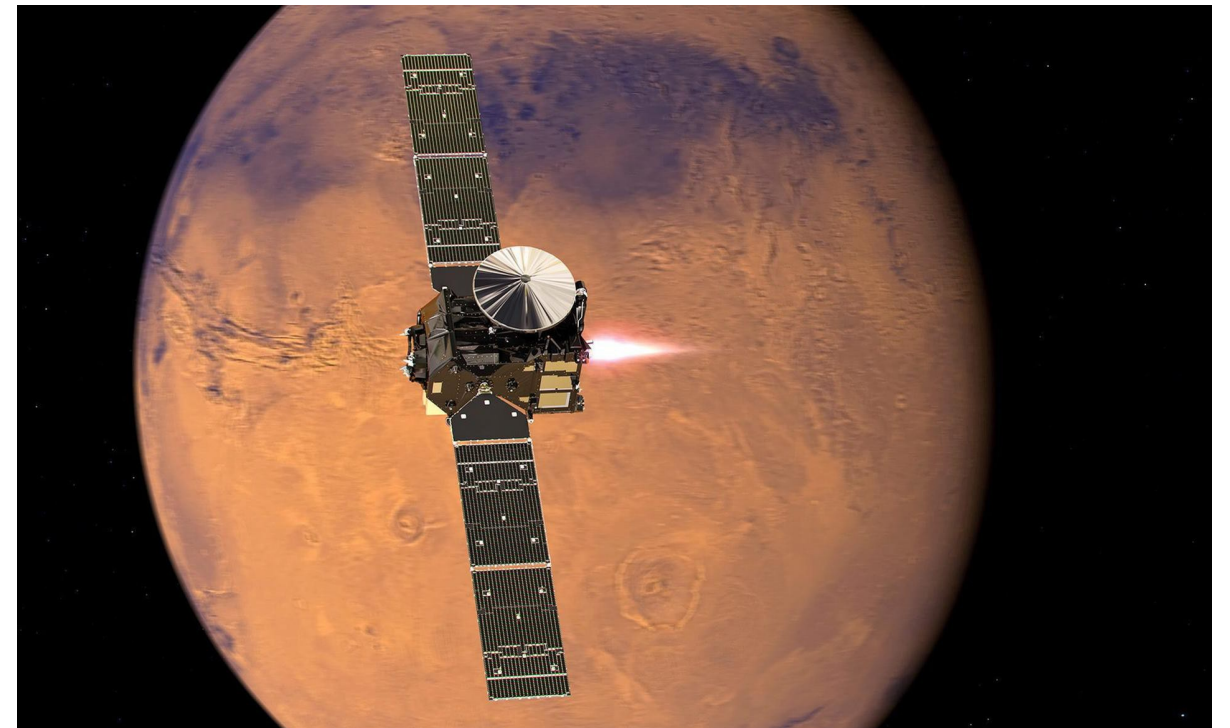
Supervisor: Jack Ogaja

13.09.2022



Presentation Outline

- Motivation
- Problem description: N-Body Solarsystem and Physics
- Approach
 - ▶ Numerical Setup and Initialization
 - ▶ Sequentiel Implementation
 - ▶ Parallel Implementation
- Performance analysis
- Conclusion



[<https://www.deutschlandfunk.de>]

Motivation

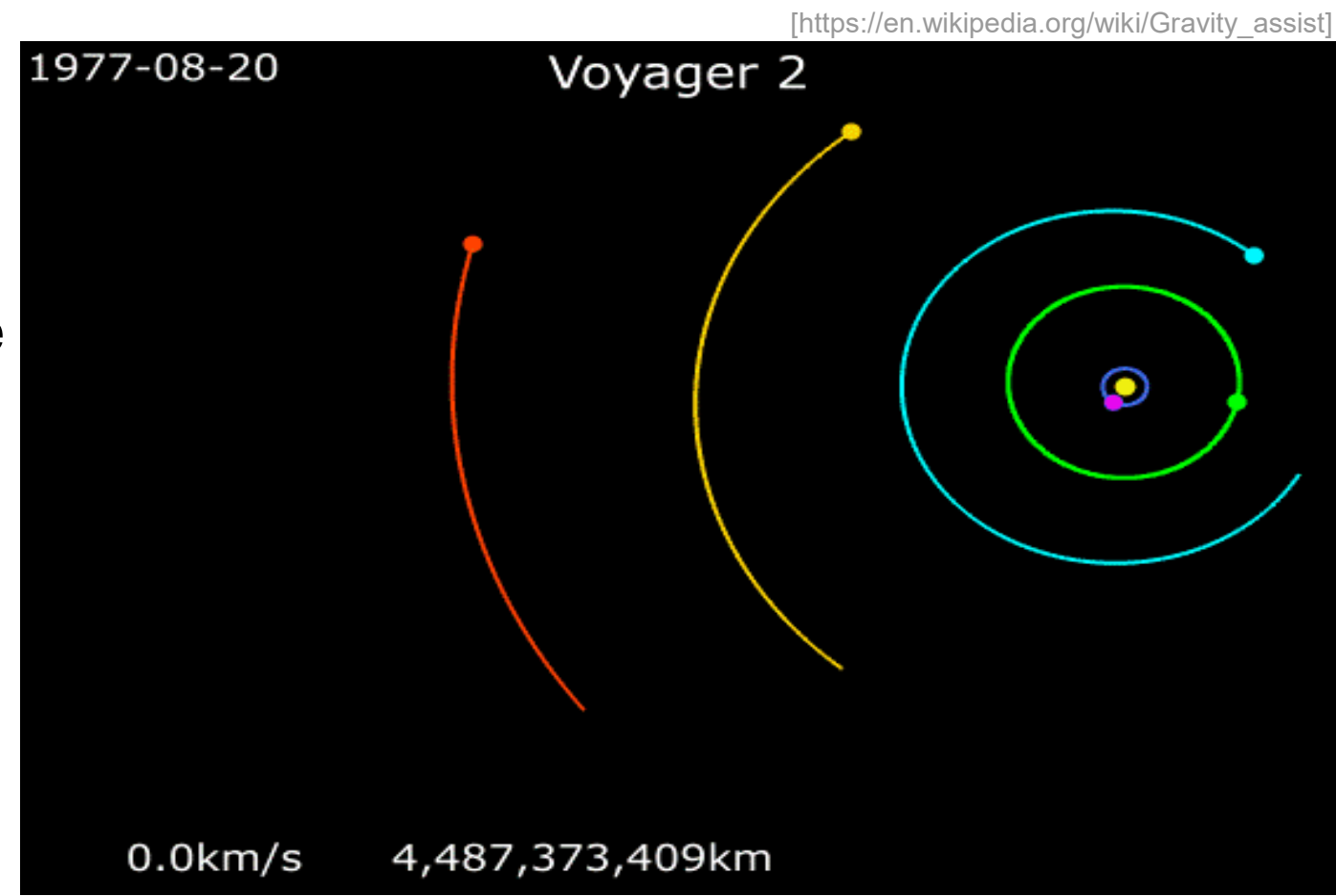
- project Idea is inspired by the Voyager program [VP]
- one of the most successful programs conducted by NASA
- goal of the VP:

- Observations and studies of the outer planets of our solar system
- Observations of the interstellar space

↳ needs to reach the Solar escape velocity in order to leave the heliosphere

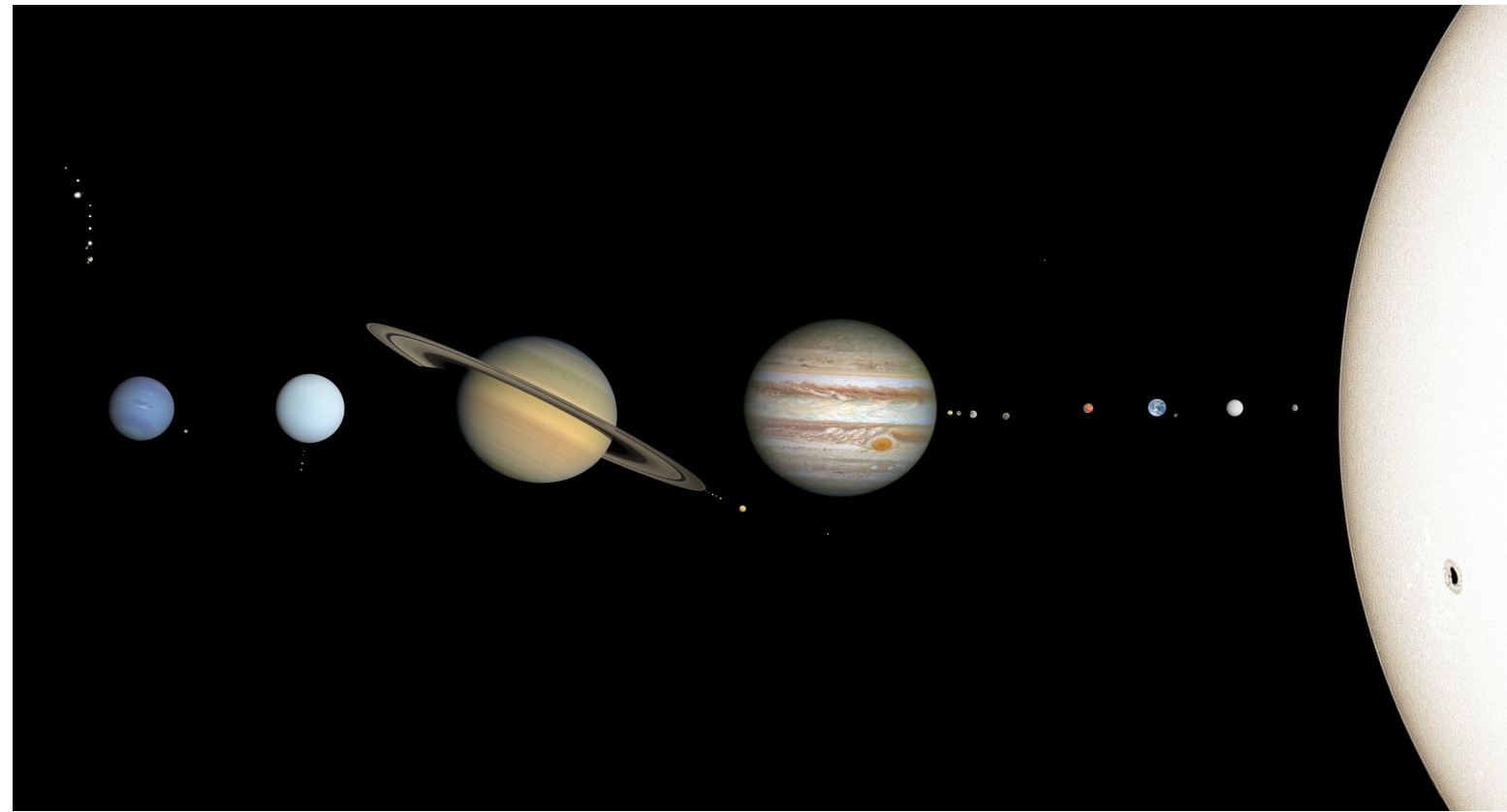
- periodic alignment of the outer planets in 1970s as foundation

↳ Our goal: perform a Grand Tour maneuver with gravity assist similar to the VP



The Problem – N-Body Solar System

- Simulation of a standard model solar system
 - Sun: $M_{\text{sun}} \gg m_N$
 - „Terrestrial“ or rocky bodies
 - Gas Giants
 - Ice Giants
- problem is given by the N-Body problem of physics
 - ↳ gravitational interaction of the N-Bodies with each other



The Problem – Physics

- gravitational force between two bodies:

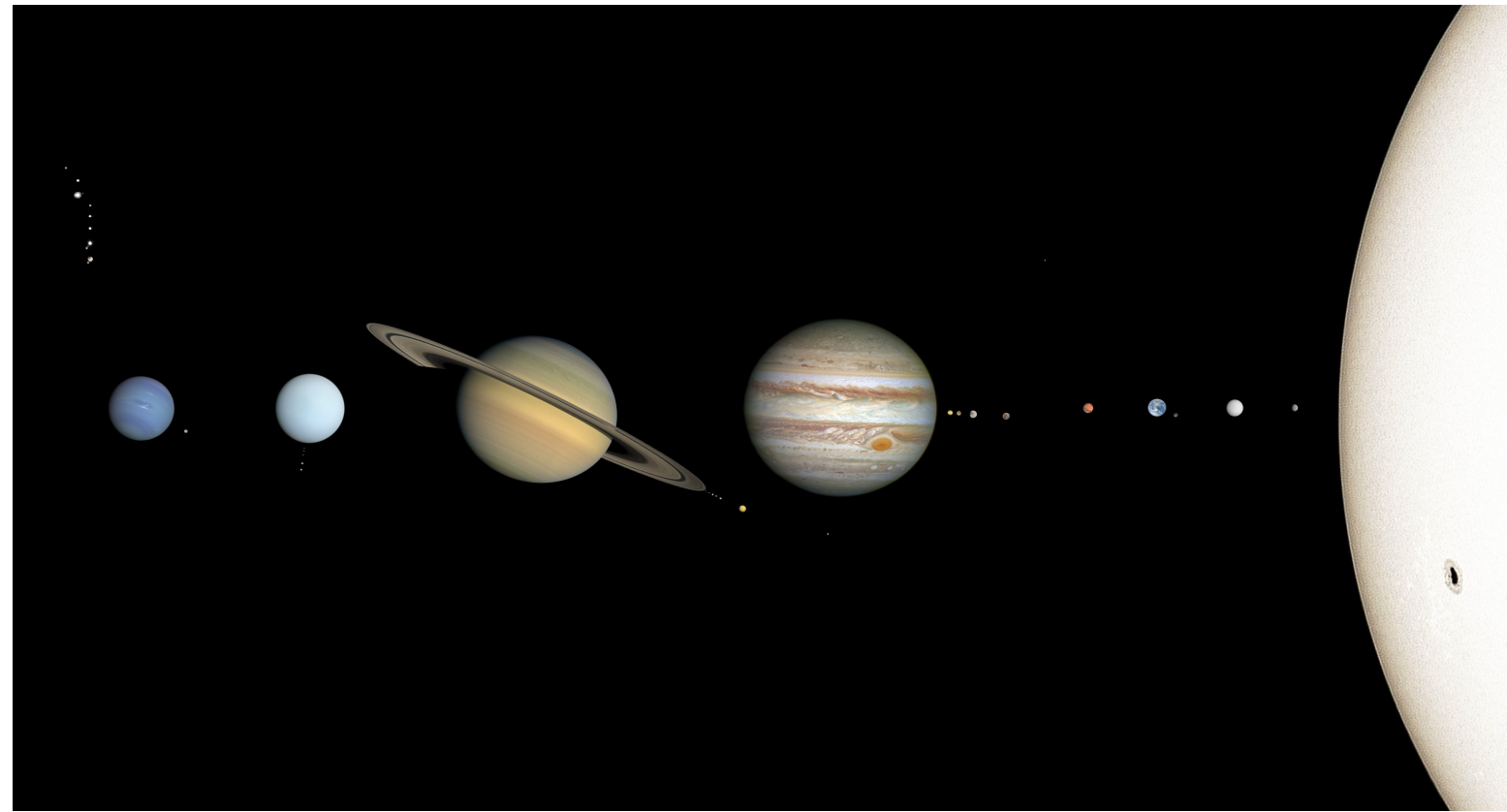
$$F_{ij} = \frac{Gm_i m_j}{d^2} \quad \text{with} \quad d = \sqrt{(x_i - x_j)^2 + (y_i - y_j)^2 + (z_i - z_j)^2}$$

- using Newton's second law: $F = m \cdot a = m_i \cdot \frac{d^2 r_i}{dt^2}$

- interaction of the N-bodies as:

$$m_i \frac{d^2 r_i}{dt^2} = \sum_{j=1, j \neq i}^N \frac{Gm_i m_j}{d^2}$$

- N^2 -interactions



The Problem – Physics

- Interaction of the N-bodies as:

$$m_i \frac{d^2 r_i}{dt^2} = \sum_{i=1, i \neq j}^N \frac{G m_i m_j}{d^2}$$

- use Lagrangian of the system:

$$\mathbf{L} = \frac{1}{2} \sum_{i=1, i \neq j}^N m_i \cdot \|\dot{r}_i\|^2 - \sum_{i=1, i \neq j}^N \frac{G m_i m_j}{d^2} \quad \text{using the momentum: } p_i = m_i \cdot \frac{dr_i}{dt}$$

The Problem – Physics

- Interaction of the N-bodies as:

$$m_i \frac{d^2 r_i}{dt^2} = \sum_{i=1, i \neq j}^N \frac{G m_i m_j}{d^2}$$

- use Lagrangian of the system:

$$\mathbf{L} = \underbrace{\frac{1}{2} \sum_{i=1, i \neq j}^N \frac{\|p_i\|^2}{m_i}}_{E_{\text{kin}} = T} - \underbrace{\sum_{i=1, i \neq j}^N \frac{G m_i m_j}{d^2}}_{E_{\text{pot}} = U} \longrightarrow \mathbf{H} = T + U$$

The Problem – Physics

- Interaction of the N-bodies as:

$$m_i \frac{d^2 r_i}{dt^2} = \sum_{i=1, i \neq j}^N \frac{G m_i m_j}{d^2}$$

- use Lagrangian of the system:

$$\mathbf{L} = \underbrace{\frac{1}{2} \sum_{i=1, i \neq j}^N \frac{\|p_i\|^2}{m_i}}_{E_{\text{kin}} = T} - \underbrace{\sum_{i=1, i \neq j}^N \frac{G m_i m_j}{d^2}}_{E_{\text{pot}} = U} \longrightarrow H = T + U$$

- Hamiltons equations of motion:

$$\frac{dr_i}{dt} = \frac{\partial H}{\partial p_i} \quad \text{and} \quad \frac{dp_i}{dt} = -\frac{\partial H}{\partial r_i}$$

4N differential equations

Approach – Numerical Setup

- code is written in C++
- Initialization of System Parameters

↳ Nondimensionalization

```

21 // PHYSICAL VALUES
22 const int Nbody = 11;
23 const int NAstroids = 200;
24 const int Ntot = Nbody + NAstroids;
25 const double AU = 1.496e11; // Astronomical unit
26 const double a = 3.169e7;
27 const double M_earth = 5.972e24;
28 const double M_sun = 332948.6*M_earth;
29
30 const double G = (6.674e-11)*M_earth*a*a/(AU*AU*AU); // Gravitational constant
31 // const double G = 6.674e-11; // Gravitational constant
32 // const double SCALE = 200.0/AU; // 1AU = 100 Pixels
33 const double TIMESTEP = 3600.0*24.0; // Timestep = 1 day

```

Astronomical Unit [AU]: distance earth-sun

Anomalistic year: Time in which earth is crossing the perihelion (nearest point so the sun)

Earth mass in [kg]

Gravitational constant in terms of Earth masses, anomalistic years and Astronomical Units

- Therefore earth velocity is given by 2π
- Object parameters are scaled with earth velocities, earth mass and AU

Approach – Initialization of Objects

- every object is defined with:
 - positions
 - velocities
 - accelerations
 - radius and animation radius
 - mass
 - pixel corresponding to the scaled position
 - orbit array
 - type

```

11 typedef enum{
12
13     SUN,
14     PLANET,
15     ASTROID,
16     ROCKET
17
18 } Object_class;
19
20
21 class Object{
22     public:
23
24     const int *color;
25     const char *name;
26
27     double x;    // positions
28     double y;
29     double u;    // velocities
30     double v;
31     double ax;  // acceleration
32     double ay;
33     double radius;
34     double mass;
35
36     int pxl_x;
37     int pxl_y;
38     int anim_radius;
39     Object_class type;
40     int orbit_transit;
41
42
43     /* Creating Array for the Orbit of the Object */
44     const int orbit_size = 4000;
45     double orbit_x[4000]; // array with size 2000 to draw the trajectory [x]
46     double orbit_y[4000]; // array with size 2000 to draw the trajectory [y]
47
48
49
50
51     void create();
52     void init(const char *name, double x0, double y0, double u0,
53             double v0, double r, double m, const int *col, int anim_r, Object_class init_type);
54
55
56     void destroy();
57
58 };

```

Approach – Initialization of Objects

- initialization by specifying desired parameters or initialization function:

```

128 for(int i = Np; i < Na; i++){
129
130     //create random angle:
131     theta = rand_val(0, 360);
132
133     //create random dist:
134     dist = mean_dist + rand_val(-delta_dist, delta_dist);
135
136     //create x and y position for dist+angle:
137     x = sin(theta)*dist;
138     y = cos(theta)*dist;
139
140     //create corresponding mean velocity:
141     mean_vel = sqrt((G*M_sun/M_earth)/(dist));
142
143     //add noise on mean velocity:
144     vel = mean_vel + rand_val(-delta_vel, delta_vel);
145
146     //create x and y velocities:
147     vel_x = cos(theta)*vel;
148     vel_y = sin(theta)*vel;
149
150     //create random mass:
151     mass = mean_m + rand_val(-delta_m, delta_m);
152
153     //initialize object:
154     Objects[i].init("AST", x, y, -vel_x, vel_y, 0.0, 1.0, WHITE, 2, ASTROID);
155 }

```

```

11 typedef enum{
12
13     SUN,
14     PLANET,
15     ASTROID,
16     ROCKET
17
18 } Object_class;
19
20
21 class Object{
22     public:
23
24     const int *color;
25     const char *name;
26
27     double x; // positions
28     double y;
29     double u; // velocities
30     double v;
31     double ax; // acceleration
32     double ay;
33     double radius;
34     double mass;
35
36     int pxl_x;
37     int pxl_y;
38     int anim_radius;
39     Object_class type;
40     int orbit_transit;
41
42
43     /* Creating Array for the Orbit of the Object */
44     const int orbit_size = 4000;
45     double orbit_x[4000]; // array with size 2000 to draw the trajectory [x]
46     double orbit_y[4000]; // array with size 2000 to draw the trajectory [y]
47
48
49
50
51     void create();
52     void init(const char *name, double x0, double y0, double u0,
53             double v0, double r, double m, const int *col, int anim_r, Object_class init_type);
54
55
56     void destroy();
57
58 };

```


Approach – Initialization of Objects

- initialization by specifying desired parameters or initialization function:

Orbital velocity: $v_o \approx \sqrt{\frac{GM}{r}}$

```

128 for(int i = Np; i < Na; i++){
129
130     //create random angle:
131     theta = rand_val(0, 360);
132
133     //create random dist:
134     dist = mean_dist + rand_val(-delta_dist, delta_dist);
135
136     //create x and y position for dist+angle:
137     x = sin(theta)*dist;
138     y = cos(theta)*dist;
139
140     //create corresponding mean velocity:
141     mean_vel = sqrt((G*M_sun/M_earth)/(dist));
142
143     //add noise on mean velocity:
144     vel = mean_vel + rand_val(-delta_vel, delta_vel);
145
146     //create x and y velocities:
147     vel_x = cos(theta)*vel;
148     vel_y = sin(theta)*vel;
149
150     //create random mass:
151     mass = mean_m + rand_val(-delta_m, delta_m);
152
153     //initialize object:
154     Objects[i].init("AST", x, y, -vel_x, vel_y, 0.0, 1.0, WHITE, 2, ASTROID);
155 }

```

```

11 typedef enum{
12
13     SUN,
14     PLANET,
15     ASTROID,
16     ROCKET
17
18 } Object_class;
19
20
21 class Object{
22 public:
23
24     const int *color;
25     const char *name;
26
27     double x; // positions
28     double y;
29     double u; // velocities
30     double v;
31     double ax; // acceleration
32     double ay;
33     double radius;
34     double mass;
35
36     int pxl_x;
37     int pxl_y;
38     int anim_radius;
39     Object_class type;
40     int orbit_transit;
41
42
43     /* Creating Array for the Orbit of the Object */
44     const int orbit_size = 4000;
45     double orbit_x[4000]; // array with size 2000 to draw the trajectory [x]
46     double orbit_y[4000]; // array with size 2000 to draw the trajectory [y]
47
48
49
50
51     void create();
52     void init(const char *name, double x0, double y0, double u0,
53             double v0, double r, double m, const int *col, int anim_r, Object_class init_type);
54
55
56     void destroy();
57
58 };

```

Approach – Initialization of Objects

- initialization by specifying desired parameters or initialization function:

for rocket escape velocity: $v_e = \sqrt{\frac{2GM}{r}}$

earth escape velocity: ~ 12 km/s

```

128 for(int i = Np; i < Na; i++){
129
130
131     //create random angle:
132     theta = rand_val(0, 360);
133
134     //create random dist:
135     dist = mean_dist + rand_val(-delta_dist, delta_dist);
136
137     //create x and y position for dist+angle:
138     x = sin(theta)*dist;
139     y = cos(theta)*dist;
140
141     //create corresponding mean velocity:
142     mean_vel = sqrt((G*M_sun/M_earth)/(dist));
143
144     //add noise on mean velocity:
145     vel = mean_vel + rand_val(-delta_vel, delta_vel);
146
147     //create x and y velocities:
148     vel_x = cos(theta)*vel;
149     vel_y = sin(theta)*vel;
150
151     //create random mass:
152     mass = mean_m + rand_val(-delta_m, delta_m);
153
154     //initialize object:
155     Objects[i].init("AST", x, y, -vel_x, vel_y, 0.0, 1.0, WHITE, 2, ASTROID);
156 }

```

```

11 typedef enum{
12
13     SUN,
14     PLANET,
15     ASTROID,
16     ROCKET
17
18 } Object_class;
19
20
21 class Object{
22     public:
23
24     const int *color;
25     const char *name;
26
27     double x;    // positions
28     double y;
29     double u;    // velocities
30     double v;
31     double ax;  // acceleration
32     double ay;
33     double radius;
34     double mass;
35
36     int pxl_x;
37     int pxl_y;
38     int anim_radius;
39     Object_class type;
40     int orbit_transit;
41
42
43     /* Creating Array for the Orbit of the Object */
44     const int orbit_size = 4000;
45     double orbit_x[4000]; // array with size 2000 to draw the trajectory [x]
46     double orbit_y[4000]; // array with size 2000 to draw the trajectory [y]
47
48
49
50
51     void create();
52     void init(const char *name, double x0, double y0, double u0,
53             double v0, double r, double m, const int *col, int anim_r, Object_class init_type);
54
55
56     void destroy();
57
58 };

```

Sequential Approach – Calculate Forces

- first calculate the distances between the Object $i \in N$ and every other object in N
- calculate gravitational force via:

$$F_{ij} = \frac{Gm_i m_j}{d^2}$$

- calculate the force in x/y direction
- calculate accelerations via:

$$F = m \cdot a$$

```
188 for(int i = 0; i < Np; i++){
189     dist_x = Plt->x - Objects[i].x;
190     dist_y = Plt->y - Objects[i].y;
191     dist = sqrt(dist_x*dist_x + dist_y*dist_y);
192
193     // do not calculate force on body on itself
194     // also, because mathematically, you would divide by dist=0
195     if(strcmp(Objects[i].name, Plt->name) != 0){
196         F = -(G * Plt->mass * Objects[i].mass)/(dist*dist);
197         F_x = F * (dist_x/dist);
198         F_y = F * (dist_y/dist);
199
200         Plt->ax += F_x/Plt->mass;
201         Plt->ay += F_y/Plt->mass;
```


Sequential Approach – Update State

- Euler Scheme for temporal discretization:

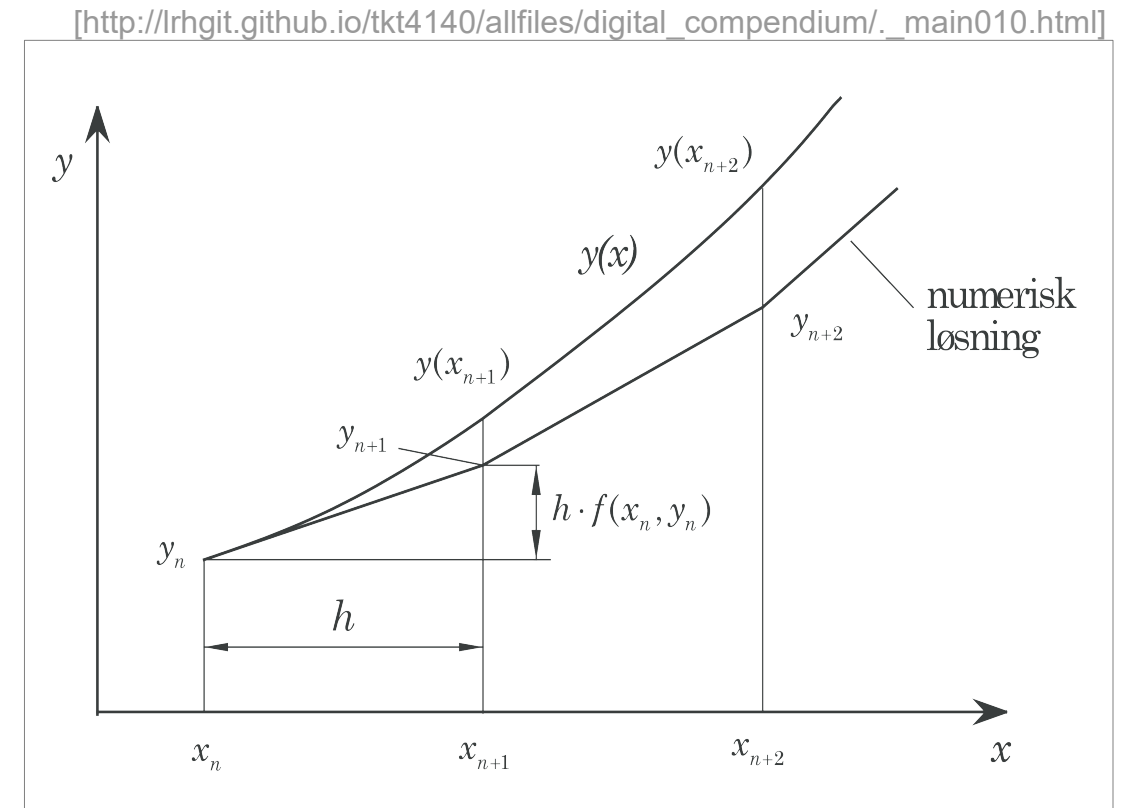
- calculate acceleration of the N-Bodies:

$$y'(x) = f(x, y(x))$$

- calculate new velocities:

$$y_{n+1}(x) = y_n + h \cdot f(x_n, y_n)$$

- do the same for positions.



```

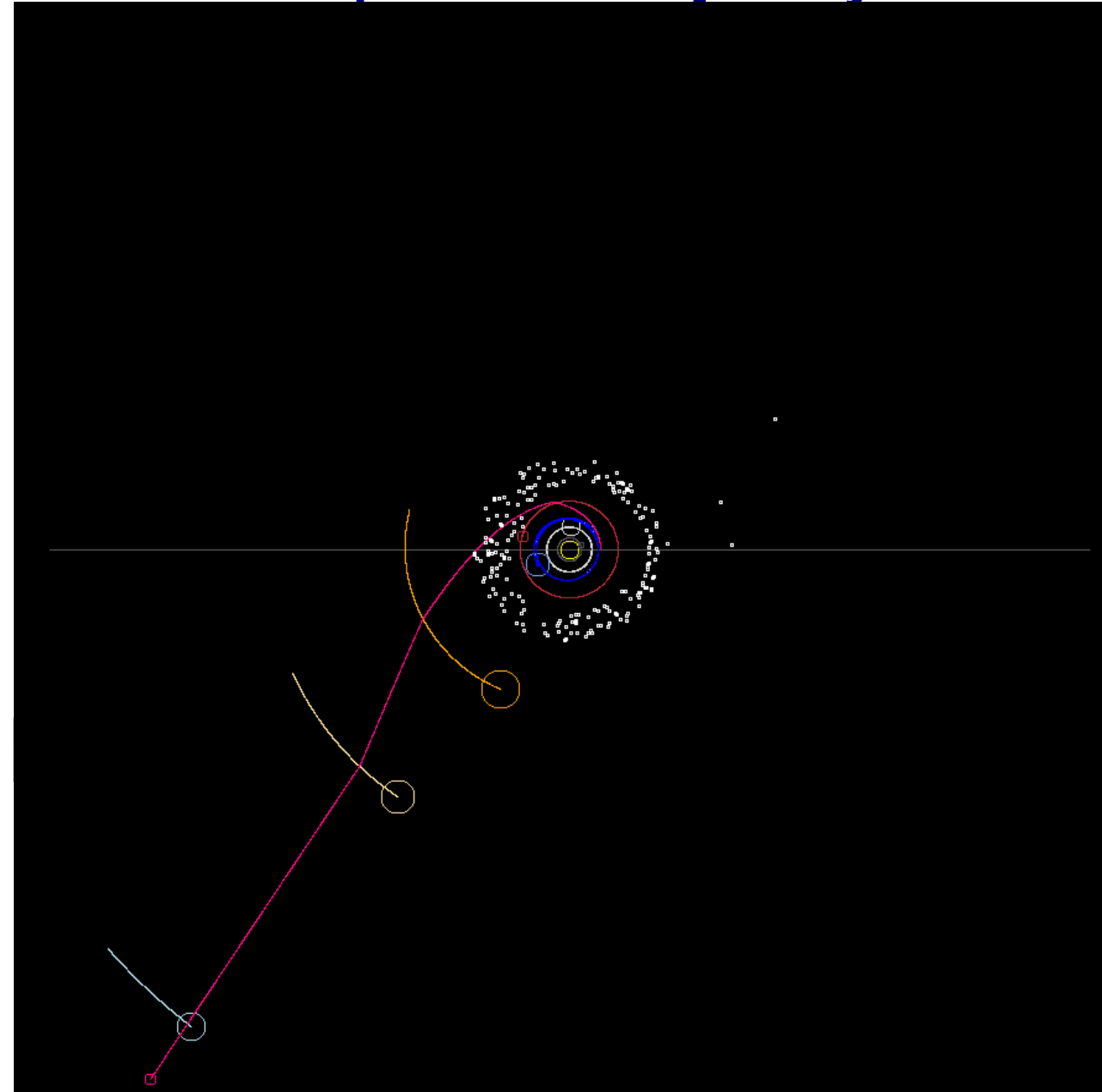
216 void System::update(Object *Plt, double delta_t, double t){
217
218     // ----- EULER step: ----- //
219     Plt->u += Plt->ax * delta_t;
220     Plt->v += Plt->ay * delta_t;
221
222     Plt->x += Plt->u * delta_t;
223     Plt->y += Plt->v * delta_t;

```

Sequential Solution

Simulation Video

Spacecraft Trajectory



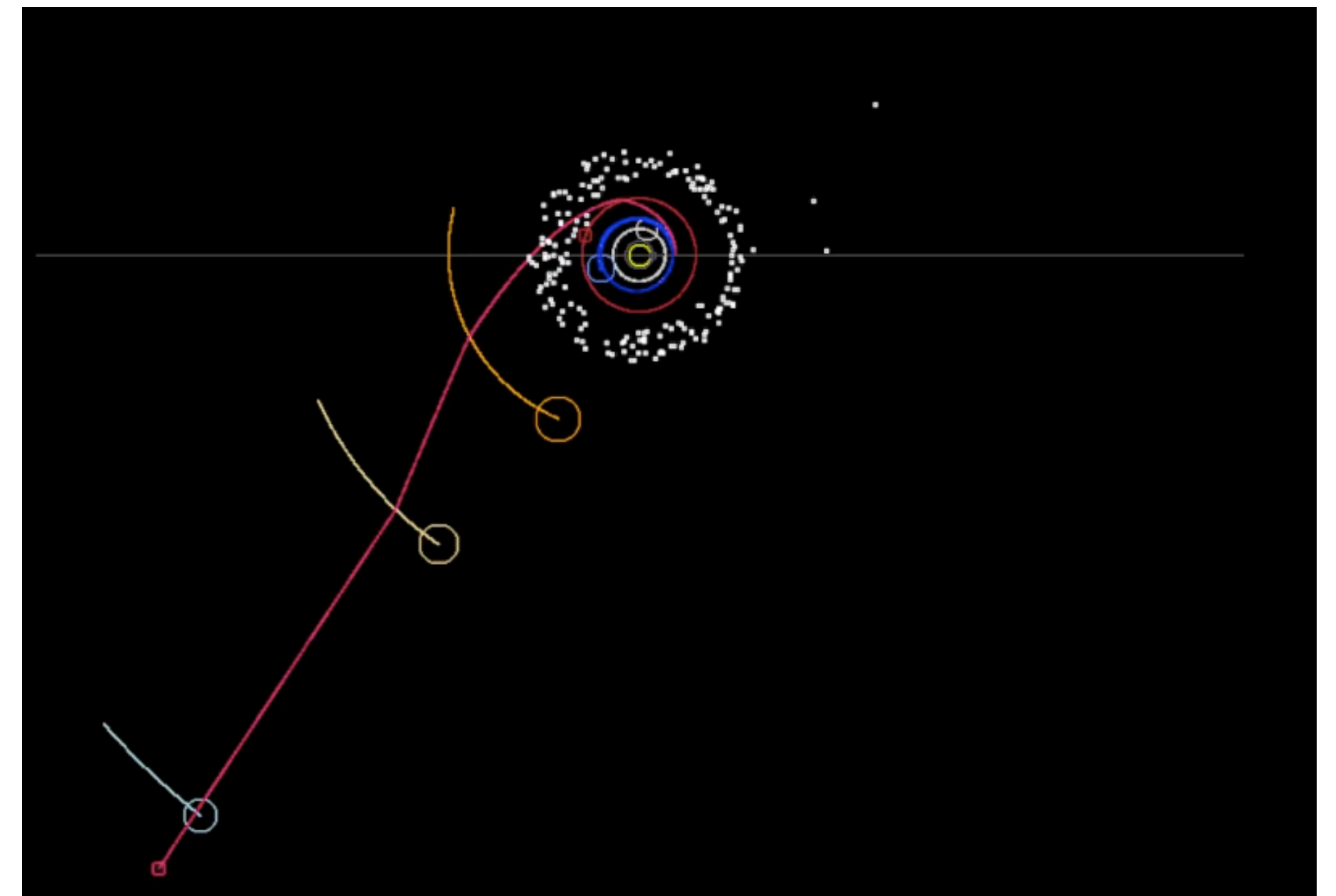


GEORG-AUGUST-UNIVERSITÄT
GÖTTINGEN

N-Body System with Swing-by

MPI Parallelization and Performance

Aaron Nagel
Yannik Feldner
13.09.2022



Parallelization

Distribution of work: Where to parallelize

Simulation Loop:

Attraction

Calculate forces
from all positions

Distribute to $P-1$ processors
to calculate $N/P-1$ forces

Collect forces from $P-1$ procs.

update

Update positions

Distribute to $P-1$ processors
to perform Euler step

Parallelization

Initialization

All Ranks:

- Initialize MPI
- Initialize own *System *sys* for allocating memory and usage of **methods**
- Allocating memory for **sending** and **receiving** data

Rank 0: coordination

Rank 1 to P-1: calculation

Simulation Loop:

|

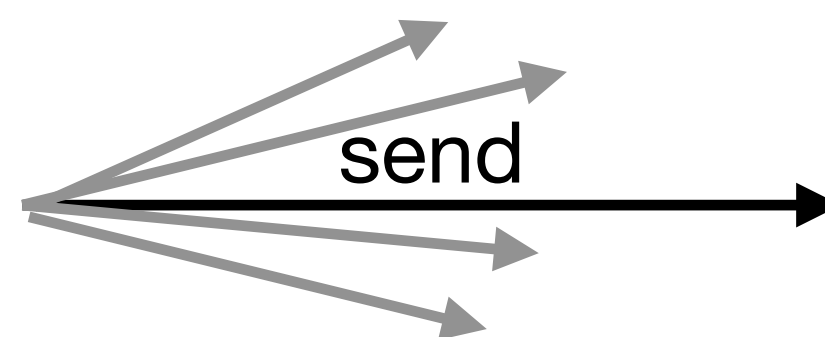
Parallelization

Distribution of data

Rank 0: coordination

Simulation Loop:

- Prepare data to send

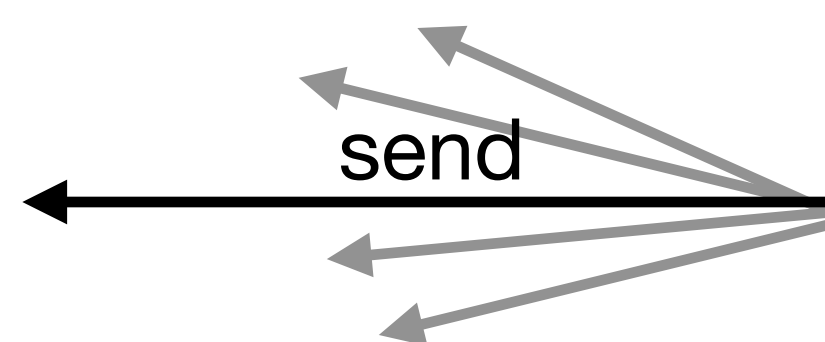


Rank 1 to P-1: calculation

- Receive **all** positions
- Sort positions in *Objects*

for $N/(P-1)$ Objects:

- `sys.attraction(sys.Objects)`
- Prepare data to send



- Receive forces
- Sort forces into system of rank 0

attraction

⋮

Parallelization

Distribution of data

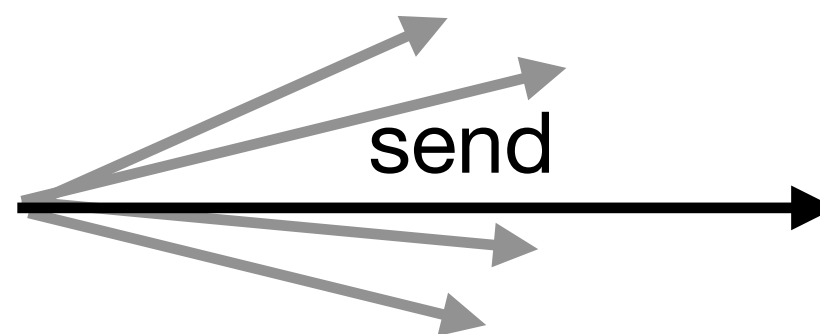
Rank 0: coordination

Rank 1 to P-1: calculation

Simulation Loop:

Update

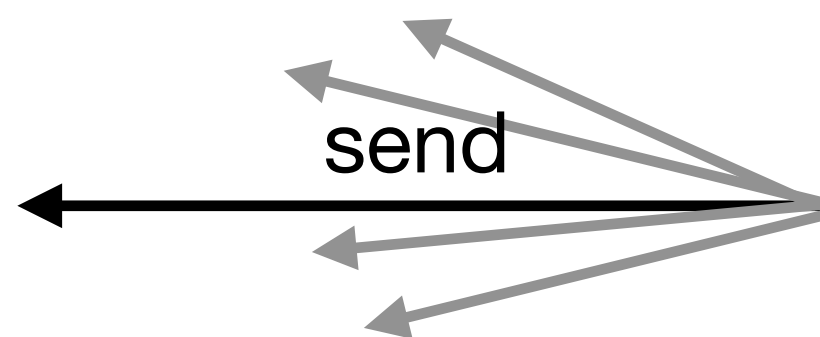
- Prepare data to send



- Receive **all** \vec{x} , \vec{v} and \vec{a}
- Sort data in *Objects*

for N/(P-1) Objects:

- Receive data
- Sort data into system of rank 0



- `sys.update(sys.Objects)`
- Prepare data to send

Parallelization

Expectation between sequential and parrallel

Sequential Solution (np = 1)

Simulation Loop:

Attraction

update

Parallel Solution on np = 2

Simulation Loop:

Proc 0

Proc 1

prep/sort data

send

send

Attraction

prep/sort data

send

update

Expectation: $t(n_p = 1) < t(n_p = 2)$

Parallelization

Work on processors 1 to P-1

```

296
297     int done = 0;
298     while(!done){
299
300         // ----- update accel: -----
301         MPI_Recv(y, 2*N, MPI_DOUBLE, 0, 0, MPI_COMM_WORLD, MPI_STATUS_IGNORE);
302         MPI_Recv(mass, N, MPI_DOUBLE, 0, 0, MPI_COMM_WORLD, MPI_STATUS_IGNORE);
303
304         // einsortieren von Zustandsvektor in system
305         for(int i = 0; i < N; i++){
306             sys.Objects[i].x = y[2*i];
307             sys.Objects[i].y = y[2*i+1];
308             sys.Objects[i].mass = mass[i];
309         }
310
311         // update accelerations using sys.attratcion Routine:
312         // only update accelerations of the N/(num_proc-1) Objects that the
313         // current processor proc. = rank is in charge:
314         for(int i = (rank-1)*N/(num_proc-1); i < rank*N/(num_proc-1); i++){
315             sys.attraction(&sys.Objects[i]);
316             // write calculated accelerations of Object i in state vector:
317             a[2*i] = sys.Objects[i].ax;
318             a[2*i + 1] = sys.Objects[i].ay;
319         }
320
321         MPI_Send(a, 2*N, MPI_DOUBLE, 0, 0, MPI_COMM_WORLD);
322

```

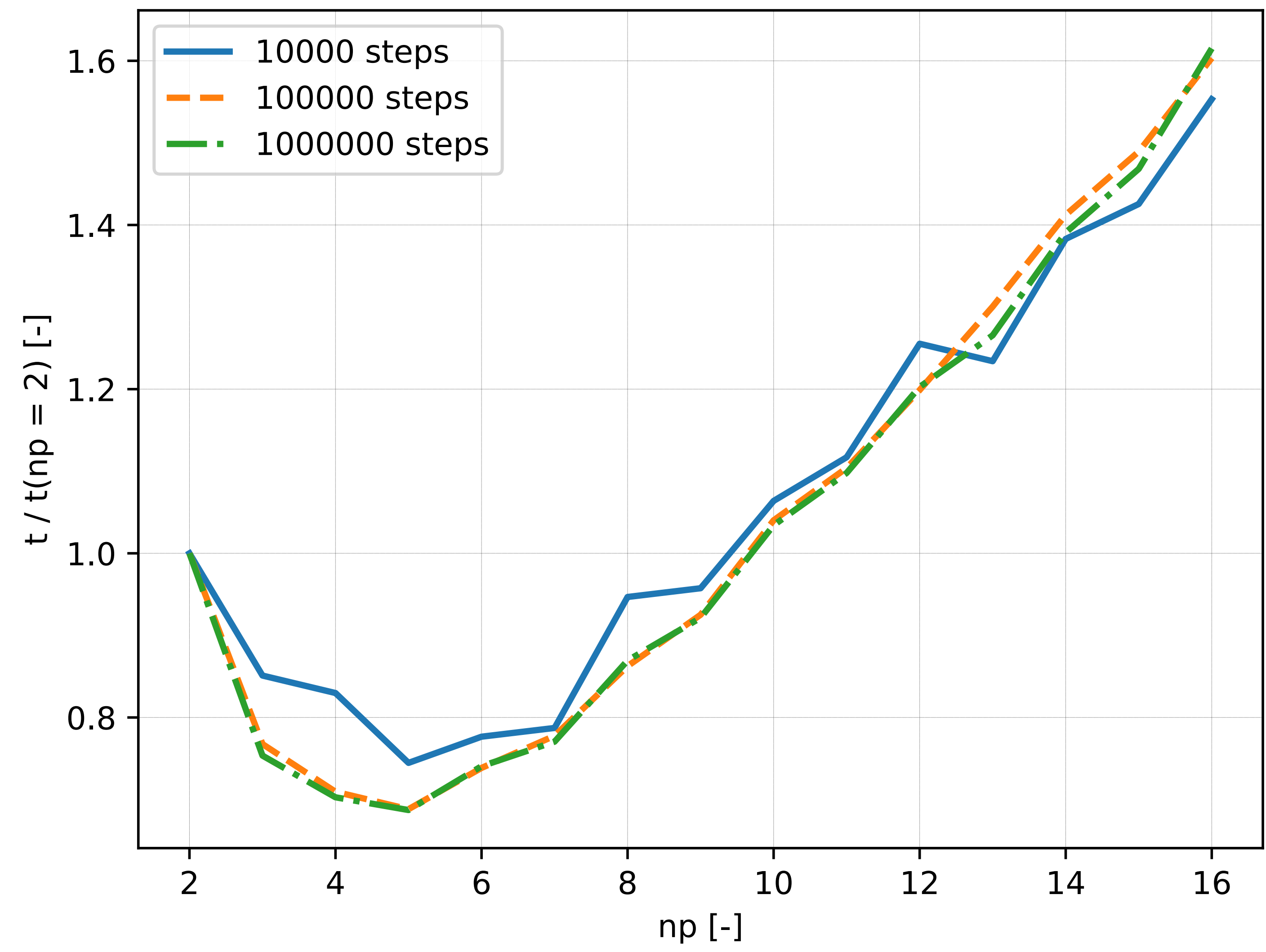
Rank 1 to P-1: calculation

- Receive **all** positions
- Sort positions in *Objects*

- **for $N/(P-1)$ Objects:**
- `sys.attraction(sys.Objects)`
- Prepare data to send

Parallelization Performance

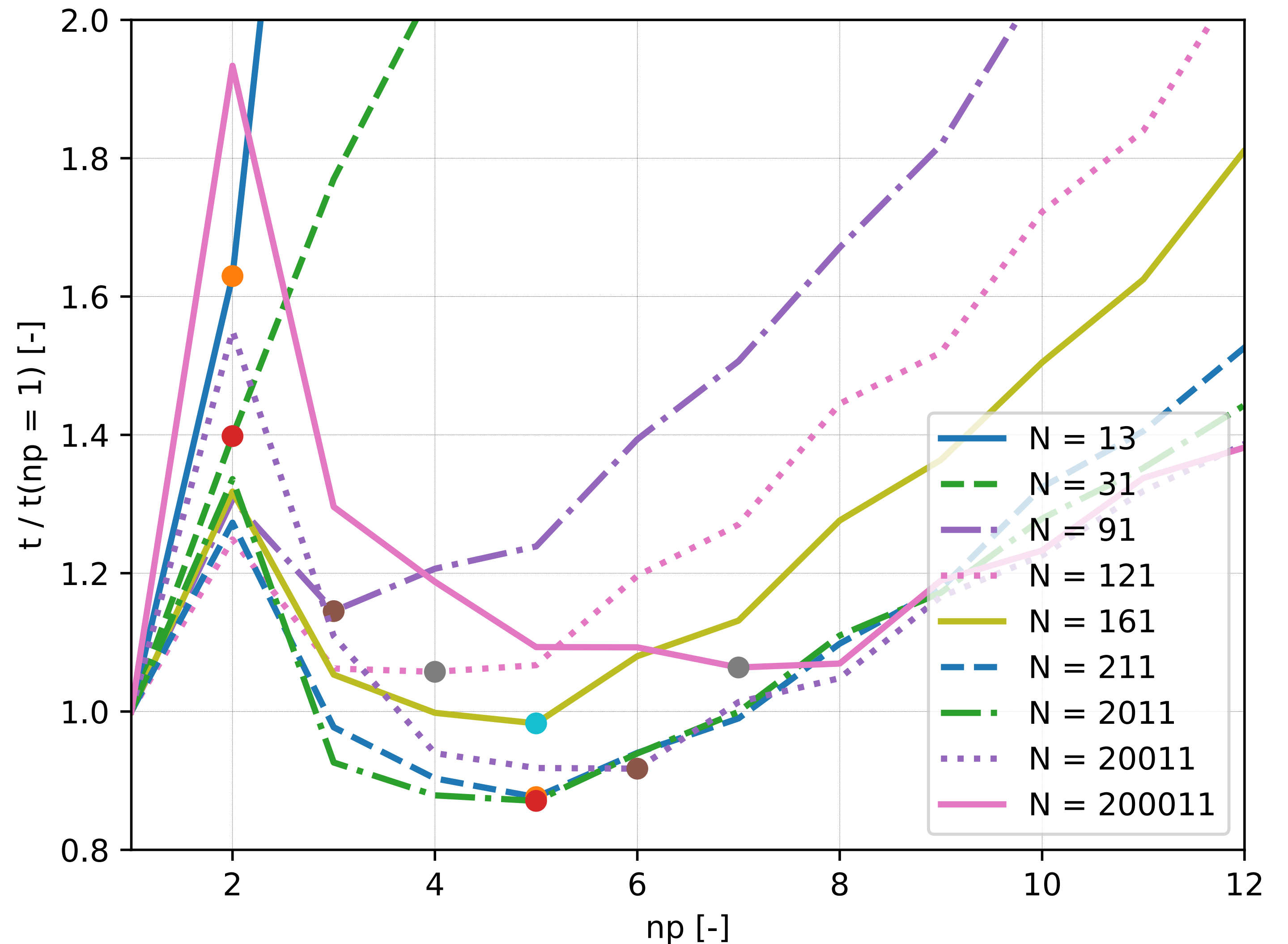
Simulation time dependence



Parallelization Performance

System size dependence

- timing without MPI_Init()
- Find balance between efficient **work load distribution** and **MPI communications**

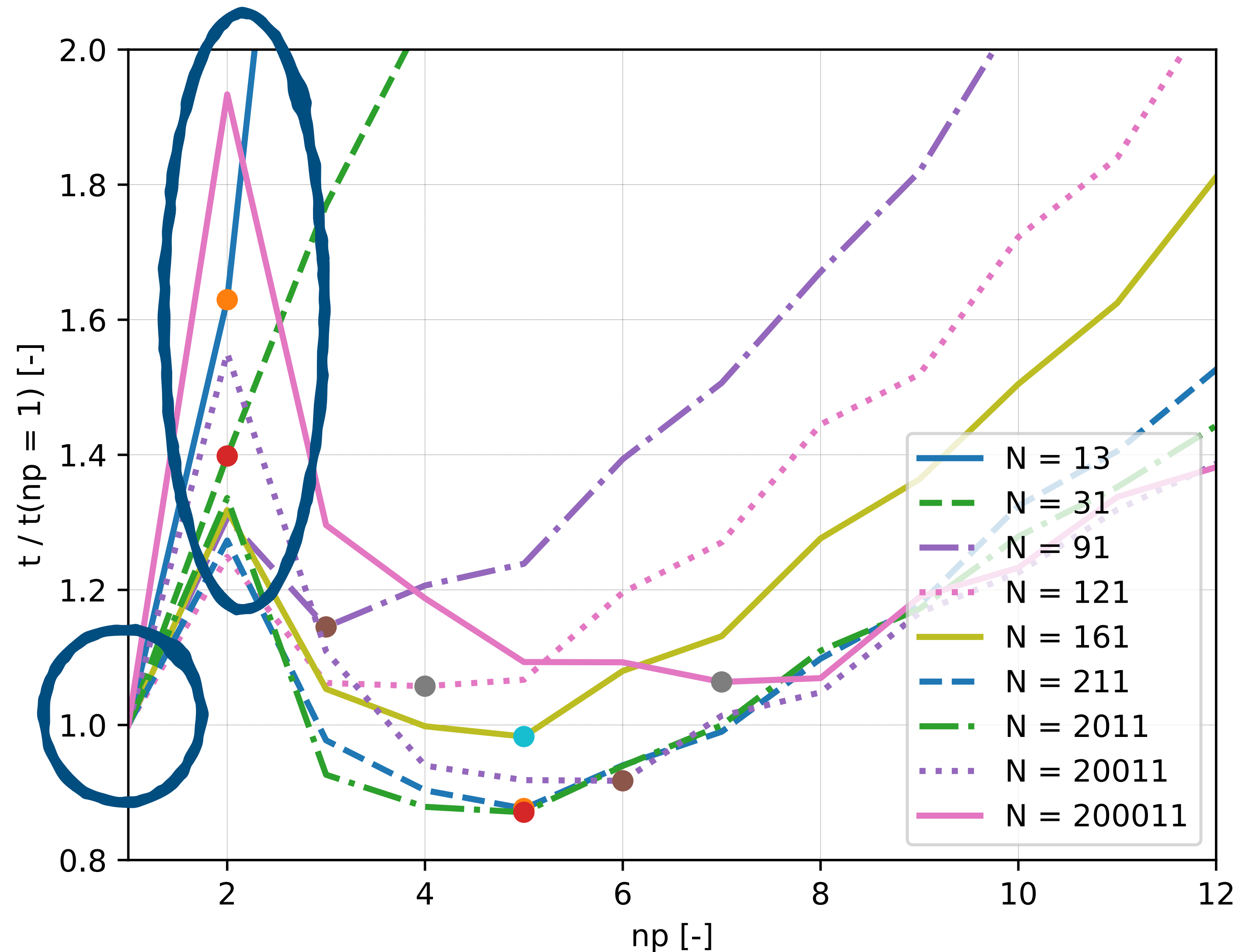


Parallelization Performance

System size dependence

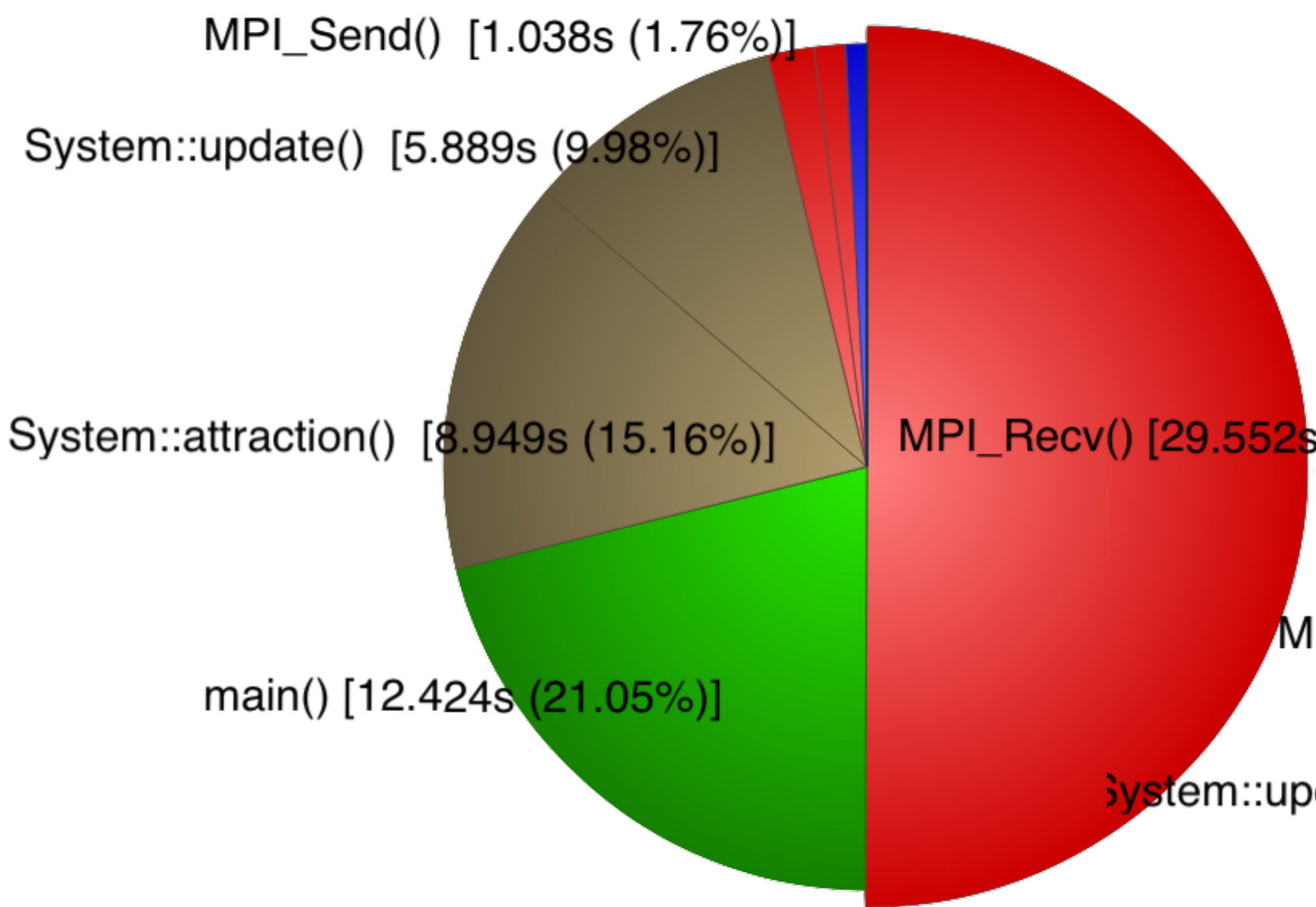
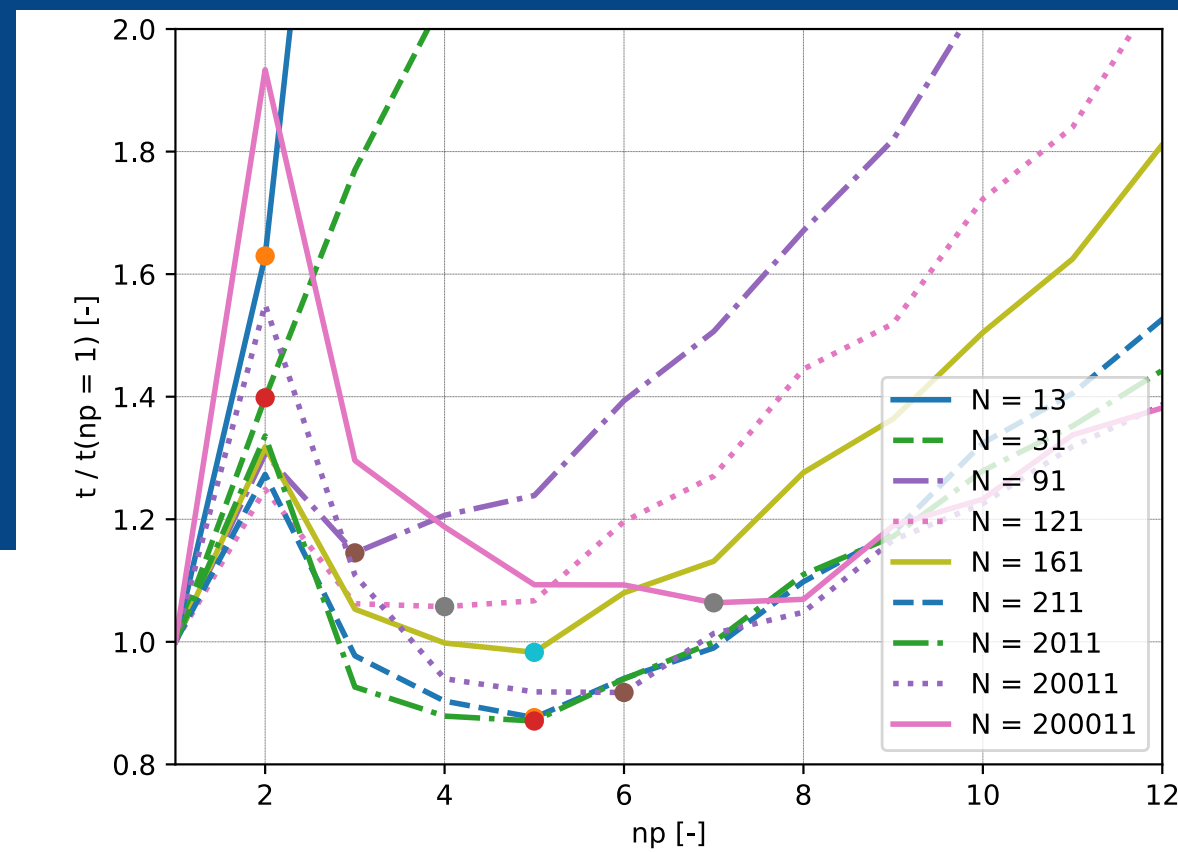
- timing without MPI_Init()
- Find balance between efficient **work load distribution** and **MPI communications**

Sequential solution ($np = 1$)
faster than parallel solution
on two procs ($np = 2$)
→ Expectation ✓

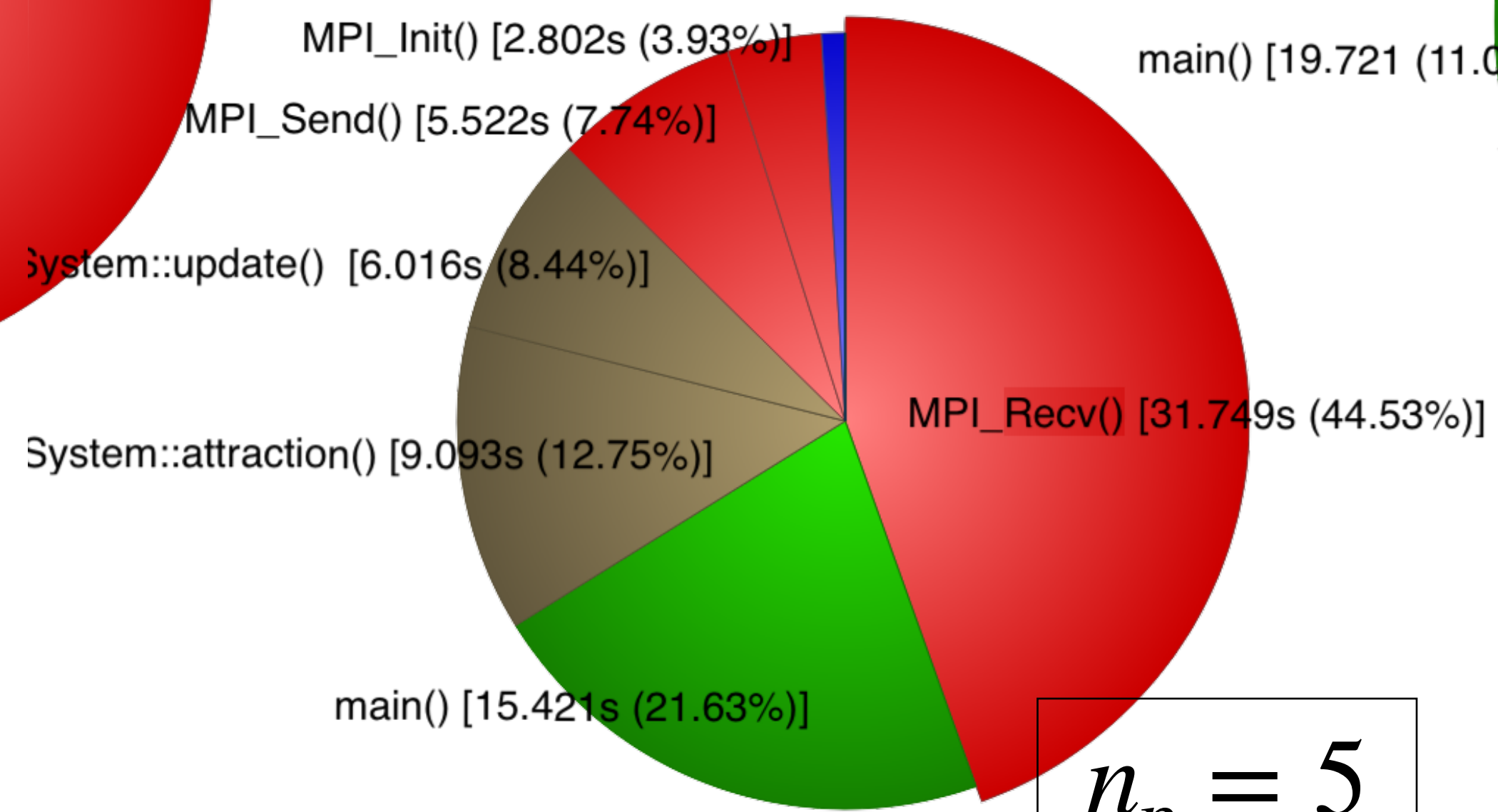


Parallelization Performance

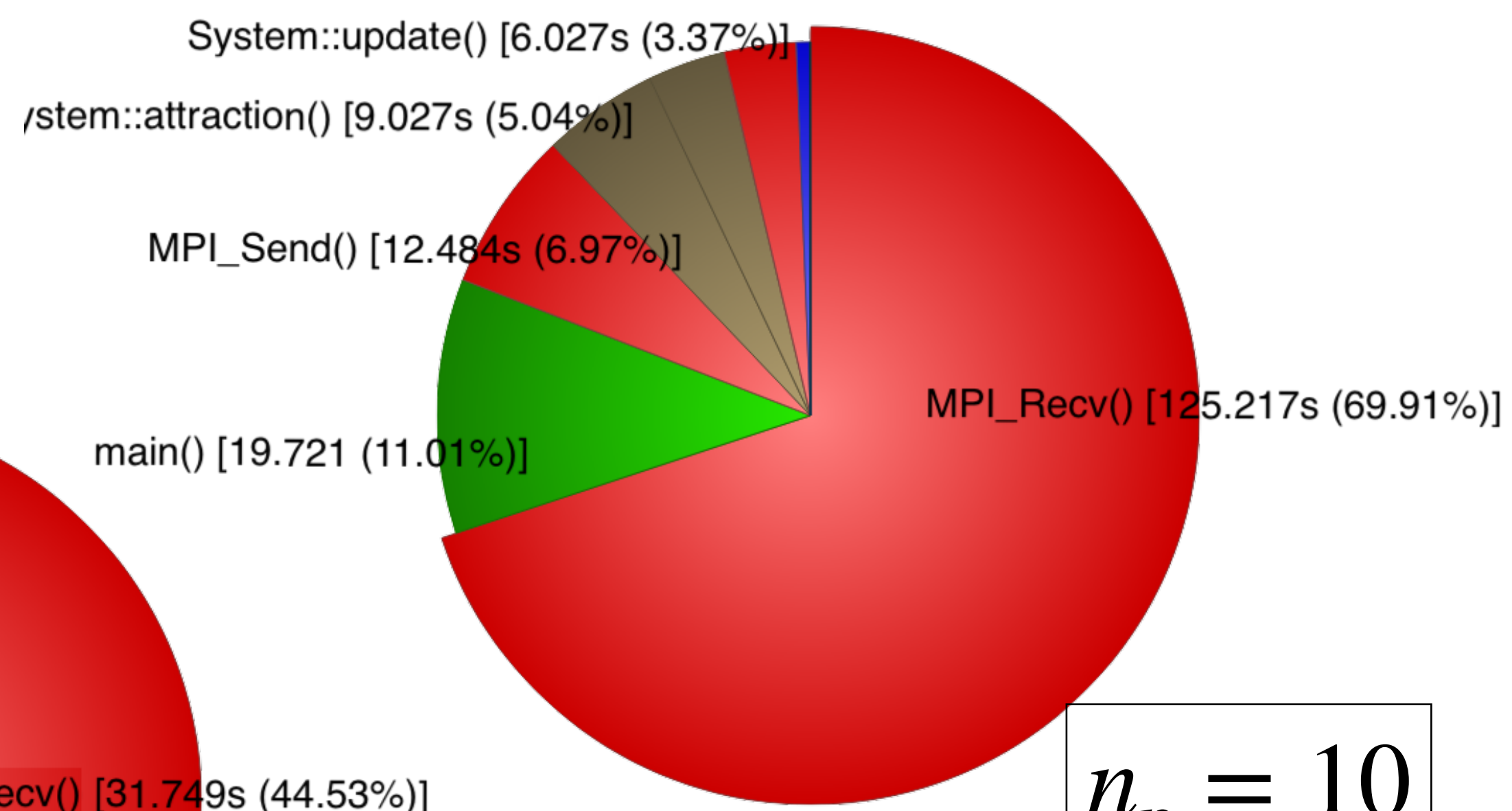
Number of Objects: $N = 211$



$n_p = 2$



$n_p = 5$

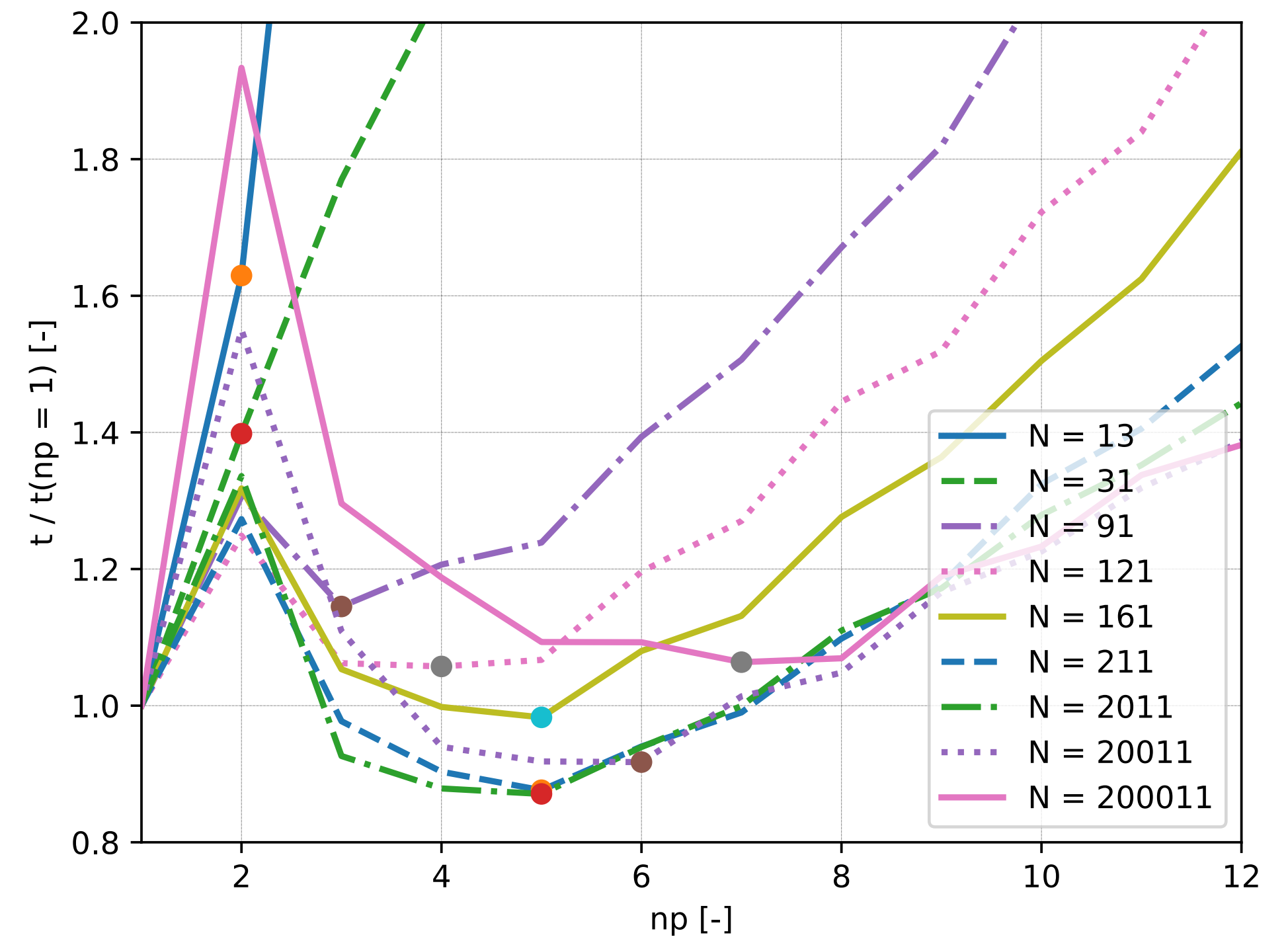


$n_p = 10$

Parallelization

Improvements

- Masses m don't change and `mass[N]` vector should be accessible by every proc. after `sys.init()` on every proc. by `sys.mass` → ~~`MPI_Send(mass, ...)`~~
- Only send necessary data for performing Euler Step on procs.
→ array length $N/(P-1)$ instead of $2N$



Parallelization

Work on processors 1 to P-1

```

308 // ----- update pos. ----- //
309 // receive data:
310 MPI_Recv(y, 2*N, MPI_DOUBLE, 0, 0, MPI_COMM_WORLD, MPI_STATUS_IGNORE);
311 MPI_Recv(v, 2*N, MPI_DOUBLE, 0, 0, MPI_COMM_WORLD, MPI_STATUS_IGNORE);
312 MPI_Recv(a, 2*N, MPI_DOUBLE, 0, 0, MPI_COMM_WORLD, MPI_STATUS_IGNORE);
313 // einsortieren von Zustandsvektor in system:
314 for(int i = 0; i < N; i++){
315     sys.Objects[i].x = y[2*i];
316     sys.Objects[i].y = y[2*i+1];
317     sys.Objects[i].u = v[2*i];
318     sys.Objects[i].v = v[2*i+1];
319     sys.Objects[i].ax = a[2*i];
320     sys.Objects[i].ay = a[2*i+1];
321 }
322 // >>> Euler Schritt
323 // perform Euler step only for the N/(num_proc-1) Objects that the
324 // current processor proc. = rank is in charge:
325 for(int i = (rank-1)*N/(num_proc-1); i < rank*N/(num_proc-1); i++){
326     // printf("Calc. attr. for Obj. %d from proc. %d \n", i, rank);
327     sys.update(&sys.Objects[i], delta_t);
328     // write calculated accelerations of Object i in state vector:
329     y[2*i] = sys.Objects[i].x;
330     y[2*i + 1] = sys.Objects[i].y;
331     v[2*i] = sys.Objects[i].u;
332     v[2*i + 1] = sys.Objects[i].v;
333     a[2*i] = sys.Objects[i].ax;
334     a[2*i + 1] = sys.Objects[i].ay;
335     pxl[2*i] = sys.Objects[i].pxl_x;
336     pxl[2*i + 1] = sys.Objects[i].pxl_y;
337 }

```

```

338
339 // Send results to proc. 0 for visualization and further coordinatiion:
340 MPI_Send(y, 2*N, MPI_DOUBLE, 0, 0, MPI_COMM_WORLD);
341 MPI_Send(v, 2*N, MPI_DOUBLE, 0, 0, MPI_COMM_WORLD);
342 MPI_Send(a, 2*N, MPI_DOUBLE, 0, 0, MPI_COMM_WORLD);
343 MPI_Send(pxl, 2*N, MPI_DOUBLE, 0, 0, MPI_COMM_WORLD);

```

send

$$i \in (p - 1)N / (P - 1)$$

$$i \in pN / (P - 1) \text{ is sufficient}$$

- Receive **all** \vec{x} , \vec{v} and \vec{a}
- Sort data in *Objects*
- **for N/(P-1) Objects:**
- sys.update(sys.Objects)
- Prepare data to send

Conclusion

- Results show exactly expected behavior
- Find balance between efficient work load distribution and MPI communications
→ minimum t in $t(n_p)$ plot
- Can **not reduce number** of Send() and Recv() calls
but the **amount of data** by reducing array length
- MPI not the best choice for N-body problems
→ e.g. shared memory approach should perform better
- **Advantage:** gained good understanding on how MPI works by N-body

